

Energy Efficient Diffuser Design For Dawt

S. Kavitha¹, R.Saravanan², M. Vairavel³

^{1,2}Assistant Professor, ³Professor, ^{1,2,3}Department of Mechanical Engineering,

¹School of Automotive and Mechanical Engineering, Kalasalingam Academy of Research and Education, Krishnankoil, Tamil Nadu, India.

² Saveetha School of Engineering, Saveetha Institute of Medical and Technical Sciences (SIMATS), Saveetha University, Tamil Nadu, India.

³ Anna Poorna Engineering College, Sankari Main Rd, Tamil Nadu, India

Abstract

In recent years, the need for renewable energies keeps on increasing. Thus, there is a need to improve the methods for harnessing the renewable energy such as wind. Wind is also sustainable, Eco friendly and zero cost energy. One such improvement is Diffuser augmented type is a class of horizontal axis Wind Turbine. The design optimization of Diffuser augmented wind turbine is discussed in this work. Along with the flangeless diffuser, the diffuser in diffuser type, flanged diffuser considered and optimized by placing of blades to maximize the velocity thereby maximize the power output. The Computational Fluid Dynamic analysis employed in the simulating flow by using Fluent of ANSYS Software. The wind inlet velocity of 5m/s and the values are compared to get optimized model to achieve good power output. From the results, the flanged Diffuser outperformed and optimized blade location was 0.135 m from the entry of diffuser and thereby a significant increase in power output determined.

Keyword:Renewable Energy, Wind turbine, optimization, computational Fluid Dynamic analysis, Diffuser.

1. Introduction

DAWT is a kind of horizontal axis class of wind turbine where blades of a rotor mounted within the Diffuser to convert the wind energy to electrical power. The arrangement of blades in DAWT, highly influences in power output. That is the blades arrangement is responsible boost up wind speed which supply by diffuser. The limitation as well as a influencing factor Betz's law by which as it is open wind turbine the maximum of 16 parts of wind energy out of 27 parts supplied by diffuser can be converted into useful work that is 59 percentage. The working phenomenon (Refer figure 1) includes that the diffuser duct provided for augmenting cross section by making the drastic pressure drop at behind the blades. Such pressure drop attracts the more flow of winds across the blade. In combination of more flow and diffuser inherent property the flow velocity increases significantly. The bare turbine model is improved and released Vortex 7 which outperformed well when compared with equal diameter of the rotor [1].

Later it was proved that the low cost energy can be produced and DAWT can be designed to work on a lower cut wind velocity and also same power out can be obtained by reducing the size of the turbine by modifying the influencing parameters to obtain same wind density. It was found that noise influencing factor in the DAWT is yaw. The yaw is usually referred in aerodynamic as misalignment between turbine pointing direction and wind direction. It is often noticed that the noise is fluctuating when wind flows accordingly, it meant that if allowed the degree of freedom the turbine align yaw as itself [2]. The figure 2 compares the conventional bare wind turbine to DAWT size reduction from the conventional wind turbine. Size reduction gives noise reduction, less hazards flying birds, mechanically lower the load and increase the life with the same output. Some of the intangible benefits like transportation savings, installation and maintenance costs, life-time improvements etc. This piece of research aimed to focus on duct design. Because [3] reported that by use of duct, one can overcome the bet's limitation of energy conversion, reduces the acoustic pollutions (by reduction of blade tip noise) and free the turbine from the attack of debris. But also the duct provides safe guard to blades from breakages [4]. [5] Explained that wind turbine operates with shroud is best solution. [6] Suggested nozzle shaped duct as well as brim shroud diffuser.

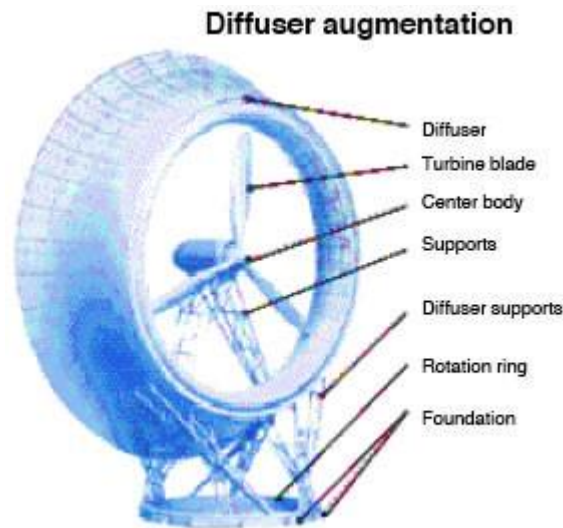


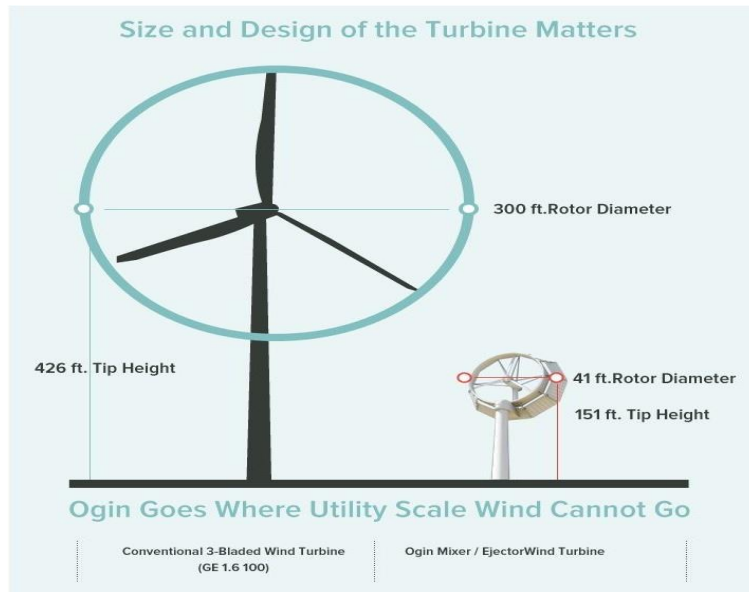
Figure 1: General Construction features of DAWT [1]

In which the area ratio was 1.67 ($A_e/A_i=1.67$) and length of the diffuser ($L=0.289$ m) was 0.289 meter for brim shroud diffuser and the area ratio was 1.19 ($A_e/A_i=1.19$) and length of the diffuser ($L=0.408$ m) was 0.408 meter for nozzle shaped duct. The nozzle shaped duct was 56% higher efficient than brim shroud diffuser due to convergent portion act as collector [7].

2. MATERIALS AND METHODS

Understanding the exact phenomenon helpsto improve its performance the figure 3 illustrate the working phenomenon of DAWT. In order to increase the power output of the DAWT, the diffuser maintains the low pressure must be at the back side of the turbine. The position of vortices formed is influencing in acceleration and deceleration of the wind in the DAWT, if vortices position is inside the diffuser which decelerated the wind, which already augmented by the diffuser will lead to low power at output of DAWT. This can be avoided by avoiding flow separation inside the diffuser and allow the vertices forms as far away from the diffuser and at the

back side.



54Figure 2: Comparing Conventional bare wind turbine and DAWT [5]

The combined effects allow the flow of wind through DAWT increase. The wind velocity decides the wind power generation, so check the possibility to augment the wind velocity. As the wind energy density is concentrated in DAWT the miniaturization was possible.

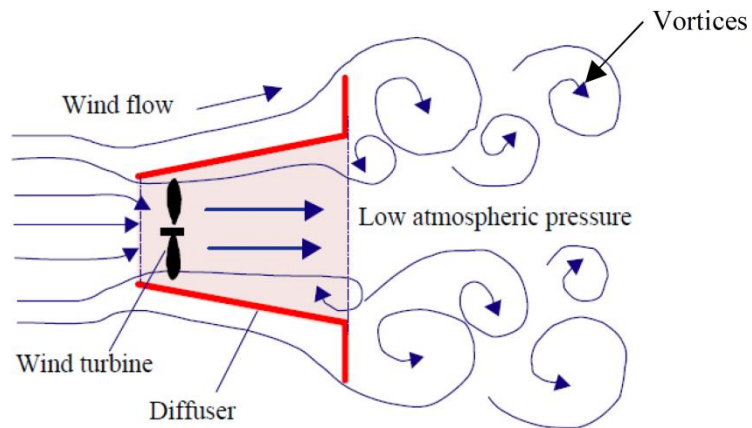


Figure 3: Principle of DAWT [8]

3. MATHEMATICAL MODEL

The one dimensional model developed to understand the behavior and optimize the parameters. The fundamental mathematical assumptions and some basic relations were derived from the earlier models of [9-13]. In the control volume the analysis assumes a control volume the ambient flow velocity of wind is U_{∞} . Similarly, wind speed at section 1-4 are: inlet of diffuser U_1 , before approaching blades U_2 , after passing over blades U_3 and exit of the diffuser U_4 . The velocity of the wind at the freestream region or inlet of diffuser are same $U_{\infty} = U_1$, within the control volume

- Number of blades not limited
- A non-rotating wake is considered
- Wind flow is in a unique direction
- Flow is steady and frictionless
- The thrust is uniform over the rotor
- The wind inside the Control volume is inviscid and homogeneous.

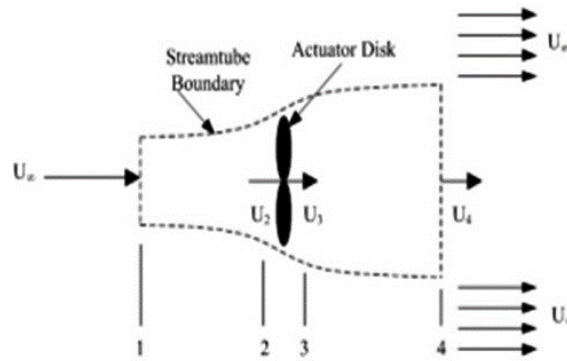


Figure 4: Modelling of DAWT

$$F_{\tau} = m(U_1 - U_4) \quad (1)$$

- Here the mass flow rate of the wind is m , U_1 , and U_4 are wind velocity at entry and the exit of the diffuser respectively. The equation (1) reveals that the velocity status that exit velocity of the wind is less than the inlet velocity of the wind. According to continuity equation for the same air density ρ and respective area of the Section A_1 and A_4 , the mass flow rate is

$$m = \rho A_1 U_1 = \rho A_4 U_4 \quad (2)$$

If there is no moving part between section 1 and 2 as well as section 3 and 4. Hence there is no work done, only flow energy, Hence the energy equation between section 1 and 2 is

$$P_1 + \frac{1}{2} \rho U_1^2 = P_2 + \frac{1}{2} \rho U_2^2 \quad (3)$$

The energy equation between section 3 and 4 is

$$P_3 + \frac{1}{2} \rho U_3^2 = P_4 + \frac{1}{2} \rho U_4^2 \quad (4)$$

The change of area before and after rotor (Section 2 and Section 3) is almost neglected as rotor blade requires uniform clearance with diffuser, hence $(A_2 = A_3)$. Due to pressure difference the work done happens. So it is mathematically stated that the thrust developed at rotor

$$F_{\tau} = A_2 P_2 - A_3 P_3 = A(P_2 - P_3) = \frac{1}{2} \rho A_2 (U_1^2 - U_4^2) \quad (5)$$

The equation of continuity at section 2 is

$$m = \rho A_2 U_2 \quad (6)$$

$$F = \frac{1m}{2U_2} (U_1^2 - U_4^2) \quad (7)$$

That is $\rho A_2 = m / U_2$

Now we equating thrust equation that

$$m = (U_1 - U_4) = \frac{1m}{2U_2} (U_1^2 - U_4^2)$$

We get that

$$U_2 = \frac{1}{2}(U_1 - U_4) \quad (8)$$

The equation 8 reveals that average of inlet and exit velocity of the control volume is rotor plane velocity.

Due to power generation certain velocity to be expended that fraction of velocity is termed as axial induction Factor a_i (it is some occasion termed as interference) factor

$$a_u = \frac{U_1 - U_2}{U_1} \quad (9)$$

From equation 9 it can be obtained that

$$U_2 = U_1 (1 - a_u) \quad (10)$$

And

Hence Power produced by turbine P_T

$$P_T = F_T U_2 \quad (11)$$

Substituting the F_T value from equation 5, equation 10 subsequently, the equation 13 will be obtained.

$$P_T = \frac{1}{2} \rho A_2 (U_1^2 - U_4^2) U_2$$

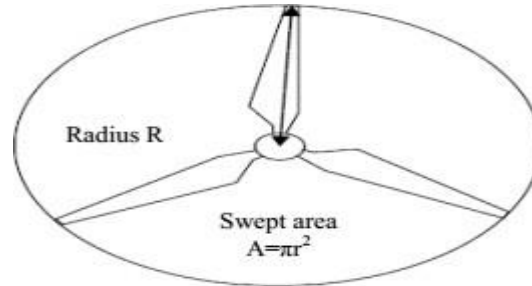
$$P_T = \frac{1}{2} \rho A_2 U_2 (U_1 - U_4)(U_1 + U_4) = \frac{1}{2} \rho A_2 U_1^3 4A a_u (1 - a_u)^2$$

Here after we consider only Rotor area A_2 is replaced by A and velocity is ambient wind velocity only so U_2 is replaced by U .

$$P_T = \frac{1}{2} \rho A U^3 4a_u (1 - a_u)^2 n \quad (13)$$

where the control volume, A_2 is replaced with A (refer Figure 5), the rotor area, and the free stream velocity U_1 is replaced by U .

Figure 5: Swept area of DWAT



The following term is called power coefficient

$$P_c = 4a_u (1 - a_u)^2 \quad (14)$$

Substituting in this in equation 13 we get

$$P_c = \frac{P_T}{\frac{1}{2}\rho AU^3}$$

Maximum possible coefficient value can be obtained by

$$= 4\left(\frac{\delta P_c}{\delta a_u} - 3a^2\right)$$

Therefore the $a_u=1/2$

Then the P_{cmax} will be.

$$P_{cmax} = 0.5926 \quad (15)$$

We know that general kinetic energy equation

$$E = \frac{1}{2} mU^2 \quad (16)$$

Here U is velocity as stated above. From the energy equation, the power is energy flow per unit time. So the change in power can be written as

$$P_T = \frac{dE}{dt} = \frac{U_2 dm}{2dt} \quad (17)$$

In which the term dm/dt is mass flow rate. The same can be written as

$$\frac{dm}{dt} = \rho A \frac{dx}{dt} \quad (18)$$

In the above equation dx/dt meant velocity U . That is $U=dx/dt$. So $dm/dt = \rho A U$. The power is

$$P_T = \frac{1}{2} \rho AU^3 \quad (19)$$

The maximum possible value of Power Coefficient (refer equation 15)

The maximum available power can be mathematically expressed as $P_{TA} = P_T P_{Cmax}$

That is

$$P_{TA} = \frac{1}{2} \rho P_{cmax} AU^3 \quad (20)$$

4. PROPOSED DESIGN

The diffuser function is basically affected by its expansion angle and length. In this design a unit length of diffuser is considered. The proportionate dimensions derived from the standard design to obtain the marginal results. Three different diffuser designs are proposed to augment the wind speed and thereby improving the DAWT output. [8] examined and declared that 1.7 times wind speed can be augmented by altering above said dimensions. [14] examined by simulation and confirmed that results. The successful dimensions were preferred in this investigation like the length of the diffuser, exit diameter and flange height.

4.1. Flangeless Diffuser

The flangeless diffuser name itself implied that there is no flange provision. The design is illustrated in the figure 6. The same size of flanged diffuser is considered for comparative study. That is the diffuser is a meter long, and its entrance diameter is 0.4 meter and expansion angle is α is 12° .

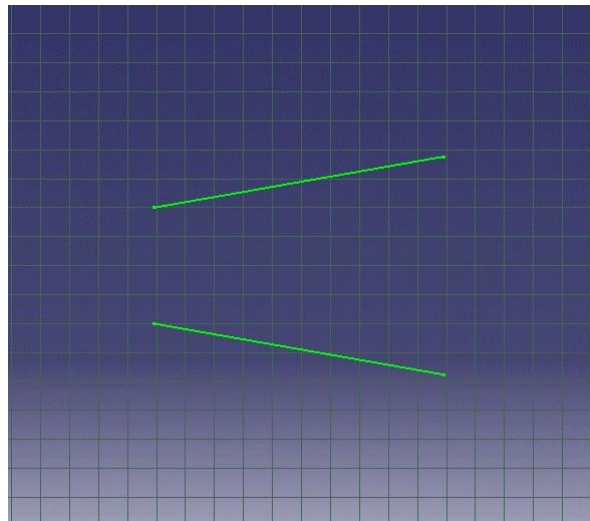


Figure 6: Flangeless Diffuser

4.2. Flangeless Diffuser in Diffuser

The flangeless diffuser in diffuser type proposed design is illustrated in the figure 7. The name flangeless implied that there is no flange provided here. The name diffuser in diffuser explained that their axis are unique. Hence the necessary dimensions are: as above said length of the diffuser (major or outer diffuser) is unit meter. The diameter and length of the inner diffuser is 50% of the outer diffuser. So as derived above the outer diffuser diameter at entrance is 0.4 meter, if so the inner diffuser entrance diameter is 0.2 meter and similarly the inner diffuser length is 0.5 meter (50% of one meter). Here the inner diffuser serves as a splitter. The purpose of the splitter is to avoid the collapse of the pressure field by shifting its vortex from inside of the diffuser. Hence the splitter diffuser keeps the vortex away from the diffuser that is on its back side of main diffuser and also avoiding the flow separation inside the diffuser. Here the expansion angle is α is 12° for diffuser, and expansion angle is α_1 for splitter is 8° .

Flanged Diffuser

The diffuser with flanged design is shown in the figure 8. As said above the length of the diffuser is unit meter. The other proportionated dimensions are the entrance diameter is 400 mm or 0.4 meter. The flange height is 200 mm or 0.2 meter. The expansion angle is α is 12° and flange height is 0.2 meter.

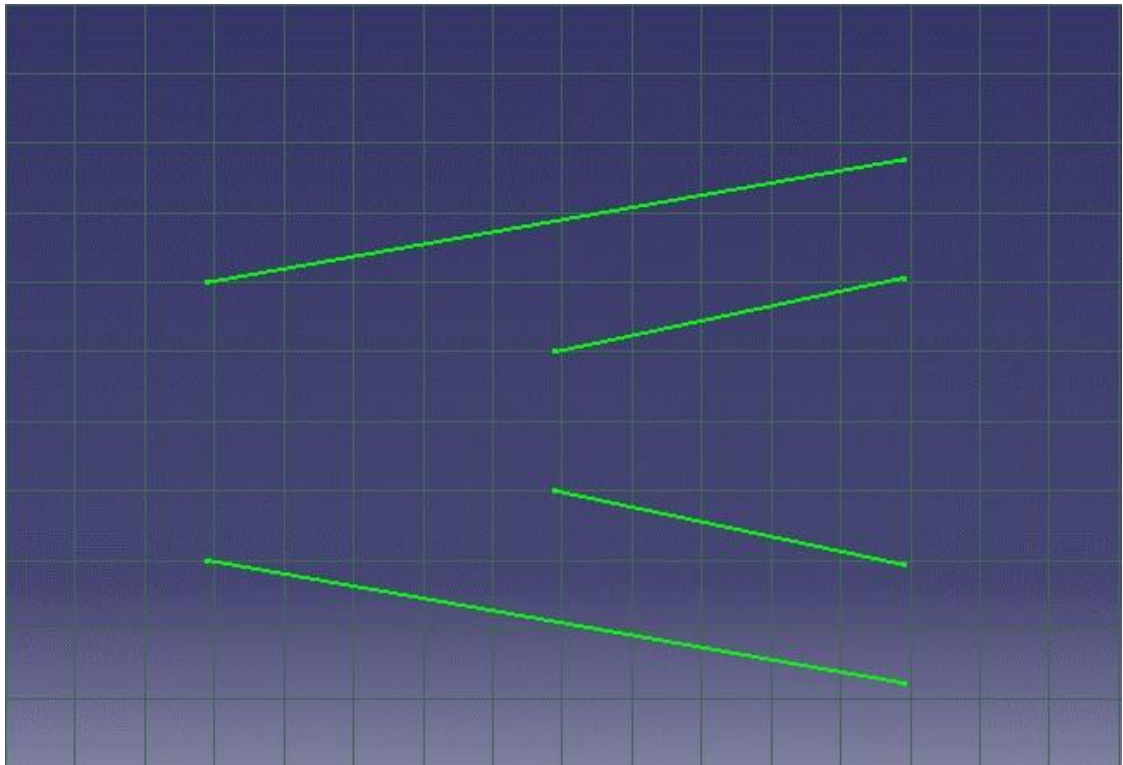


Figure 7: Flangeless Diffuser in Diffuser

4.3. BOUNDARY CONDITIONS

Figure 9 shows the sample investigation of defining boundary condition for flanged diffuser. The analysis is considered axisymmetric. The half cone angle is ϕ here the flange height h is 0.2 meter the rear end diameter is $0.5D$ that is 0.2 meter. The air inlet heights that is slip height $9.5D$. The inlet located from one side of control volume is $5D$ and $8.5D$ before the opposite side of the control volume. The wind flow assumed here unidirectional that is x direction.

4.4. FLANGELESS DIFFUSER

The flangeless diffuser in diffuser case is considered here for dynamic analysis in the ANSYS Fluent 15.0. the velocity contours magnitudes were investigated. The uniform wind speed of 5 meter per second is considered. The figure 10 demonstrates the results of the velocity distribution of flangeless diffuser in diffuser case. The maximum velocity could be augmented up to 7.28 meter per second in this proposed flangeless diffuser in diffuser case.

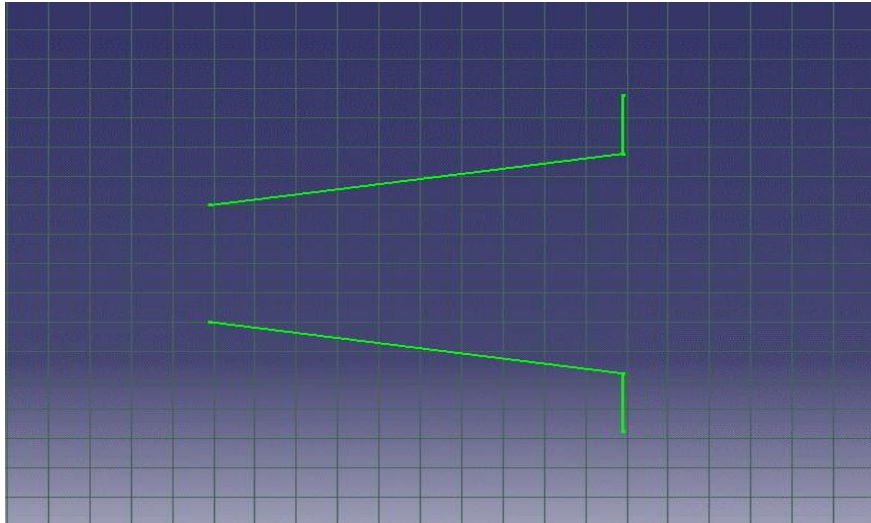


Figure 8: Flanged Diffuser

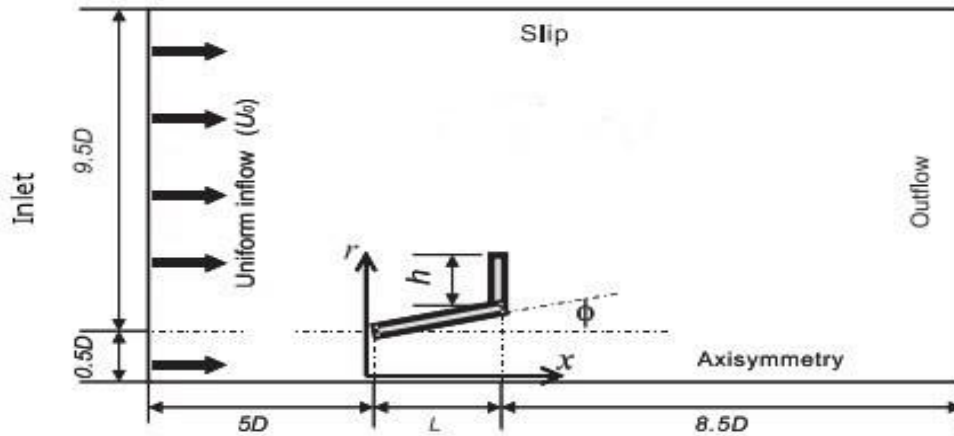


Figure 9: Flanged diffuser standard Boundary conditions

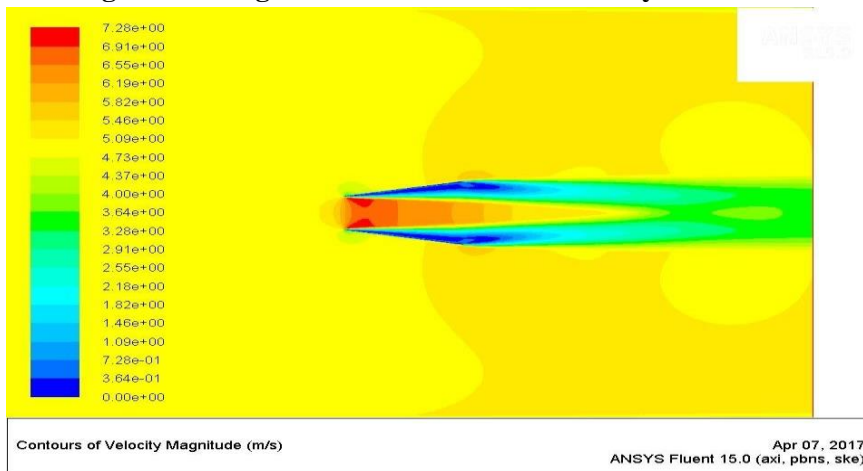


Figure 10: Velocity distribution Results of dynamic analysis on Flangeless Diffuser
 4.5. DYNAMIC ANALYSIS ON FLANGELESS DIFFUSER IN DIFFUSER

CASE

The flangeless diffuser in diffuser case is considered here for dynamic analysis in the ANSYS Fluent 15.0. the velocity contours magnitudes were investigated. The uniform wind speed of 5 meters per second is considered. The figure 11 demonstrates the results of the velocity distribution of flangeless diffuser in diffuser case. The maximum velocity could be augmented up to 6.70 meters per second in this proposed flangeless diffuser in diffuser case.

4.6. DYNAMIC ANALYSIS ON FLANGED DIFFUSER CASE

In this analysis the flanged diffuser case is considered for dynamic analysis in the ANSYS Fluent 15.0. The uniform wind speed of 5 meter per second is considered. The figure 12 exhibits the velocity distribution result of dynamic analysis on flanged diffuser for defined boundary conditions. It is noticed that the maximum augmented wind speed is 8.02 meter per second.

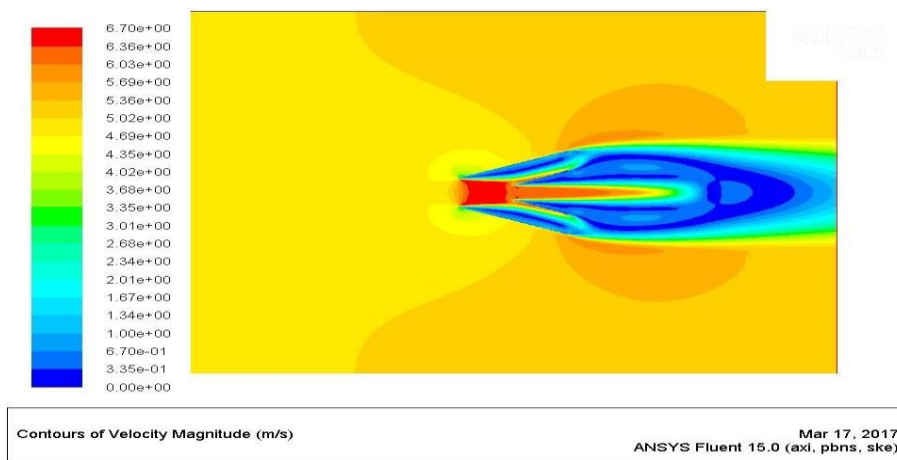


Figure 11: Velocity distribution Results of dynamic analysis on Flangeless Diffuser in Diffuser case

5. RESULTS AND DISCUSSIONS

5.1. FLANGELESS DIFFUSER

Table 1: Results of this Research

Proposed Diffuser	Maximum augmented wind speed (m/s)	Percent of wind speed Augmented	Power output Augmentation in terms of No diffuser case (P)	Best Blade Location from entrance	Comments
Flangeless Diffuser	7.26	If abbreviations are used in the text they should be defined in the text at first use, and a list of abbreviations should be provided.	Results should be clear and concise.	Title page text...	
Flangeless Diffuser	8.02	60.34%	3.53 P	0.135	Almost all layer of wing augmented so Safe and strong

The velocity profile is depicted in the figure 13 for the case of flangeless diffuser. The low pressure regions are it was ensured that the maximum velocity of 7.26 meter per second

augmented by means of flangeless diffuser. The wind expanded on through the diffuser with the expense of pressure drop the velocity is augmented. In the figure 14 it is clear that the velocity gets dropped after a certain distance in the normal direction. The normal distance is equally spitted in seven parts and analyzed the details of the study result depicted in the figure 15. It should be noticed in the figure 15 that the green and dark blue colored graph which show the status 2/7 and 3/7 part of the vertical distance. It shows the augmentation profile of wind from the entry to exit. The maximum wind speed augmentation from assumed 5 meter per second is 7.260 meter per second. This implied that the flangeless diffuser augmented wind speed by 45.20%. The blade location's distance is 0.165 meter to utilize the high speed wing. The problem is suddenly fallen of velocity after 0.16 meter distance from the entrance. The acceleration is not found similar for all layers of wind below with respect to height except 2/7 and 3/7 distance from the axis.

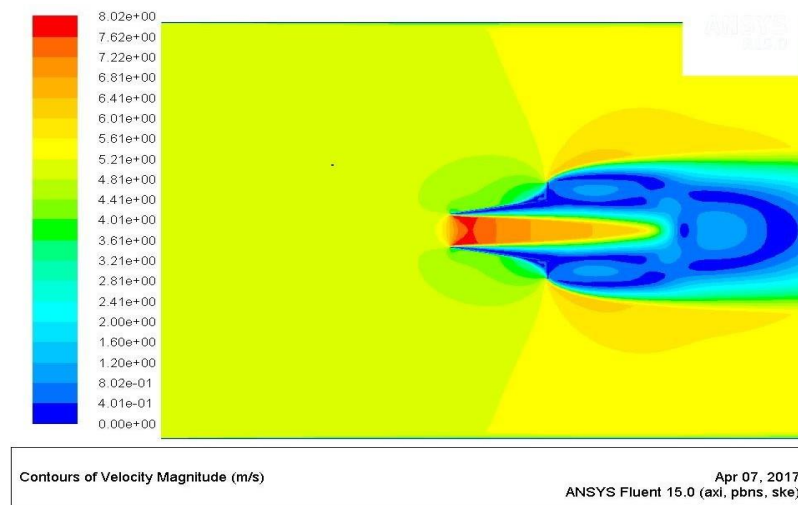


Figure 12: Velocity distribution Results of dynamic analysis on Flanged diffuser case

5.2. FLANGELESS DIFFUSER IN DIFFUSER CASE

The velocity distribution of flangeless diffuser in diffuser is shown in the figure 15. The inner diffuser and outer diffuser velocity distribution profiles show clearly in the figure 16.

The axially velocity augmentation is found good.

But the inner diffuser reduced the size of the diffuser and hence the net augmentation is diminished than the flangeless diffuser case. The velocity augmentation region continues at inner diffuser. The maximum wind speed observed that 6.696 meter per second. That is 33.99% wind speed is augmented by this flangeless diffuser and diffuser setup. The figure 16 shows that the velocity augmented by both diffusers. The dotted lines show the velocity augmentation with respect to different layers in normal directions. Similarly, the solid lines show the splitter diffuser augmentation. As expected the splitter continuously augment the wing throughout its length,

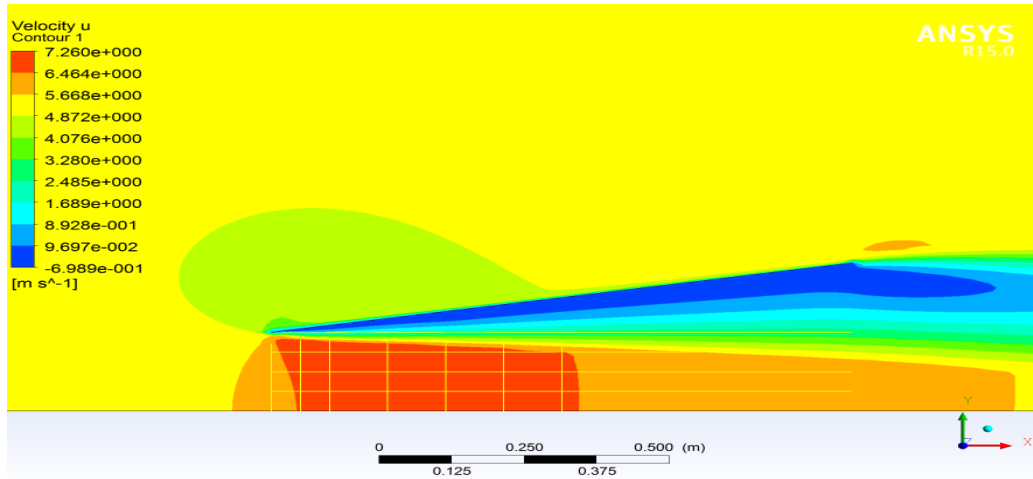


Figure 13: The Results of Linear Velocity distribution for flangeless diffuser case

This case to be enriched by suitable analysis in future studies. At 0.135-meter distance from the entrance is the optimal location of the blade in this case to utilize augmented maximum the velocity to convert the wind power to mechanical power. The wind blow at all layers, are get augmented in terms of velocity well. The pattern of all layers found uniform. The splitter diffuser is also supported well to augment the velocity of wind. Ultimately all layers the wind blow is augmented in terms of velocity. But sudden fall of velocity after 0.135 meter is noticed for all layers.

5.3. Linear Velocity results on Flanged Diffuser

The figure 17 shows the velocity distribution of flanged diffuser case. The velocity augmentation is found gradually improved as well higher than the flangeless diffuser as well as flangeless diffuser in diffuser cases. The maximum velocity of 8.017 meter per second is observed. That is a maximum of 60.34% wind speed is augmented by means of this flanged diffuser. The uniform pattern is achieved for both the flanged diffuser height only extremelayers like except 10/12, 11/12 and 12/12 augmentation is gradually improved. The flanged diffuser ensured the proper wind speed augmentation than the flangeless diffuser as well as flangeless diffuser in diffuser cases.

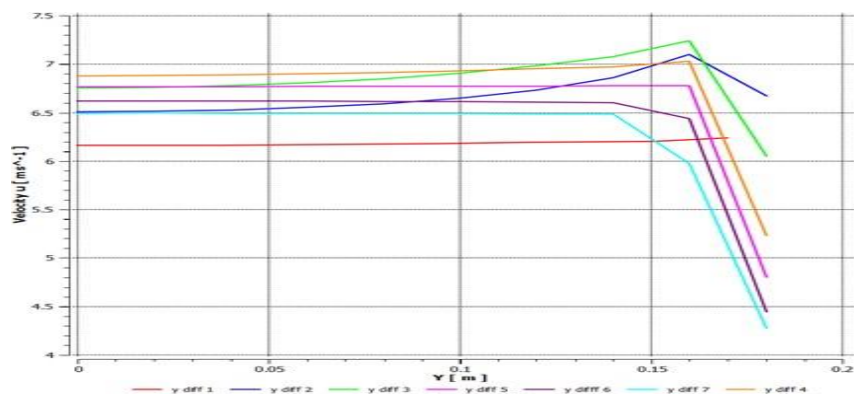


Figure 14: Linear Velocity distribution along the normal direction for flangeless diffuser case.

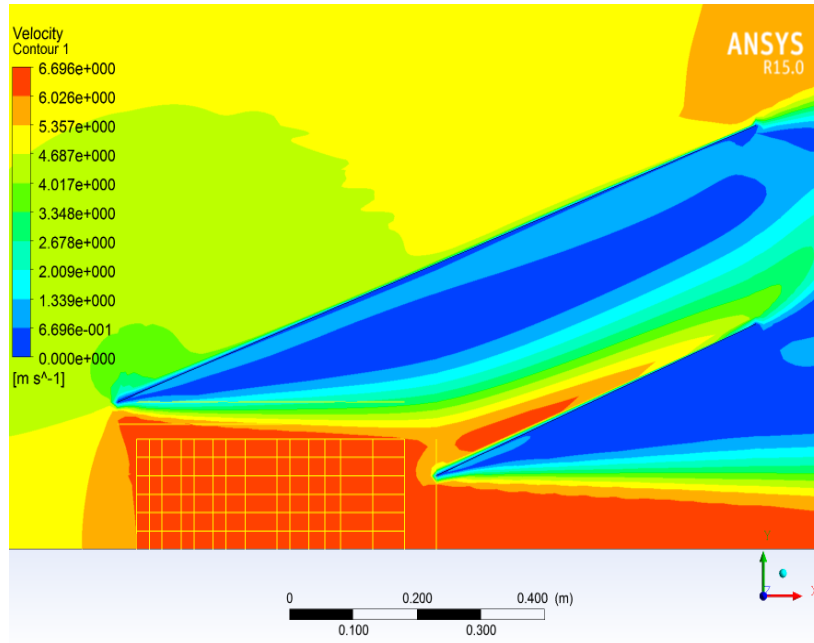


Figure 15: Results of Linear Velocity distribution for flangeless diffuser in Diffuser

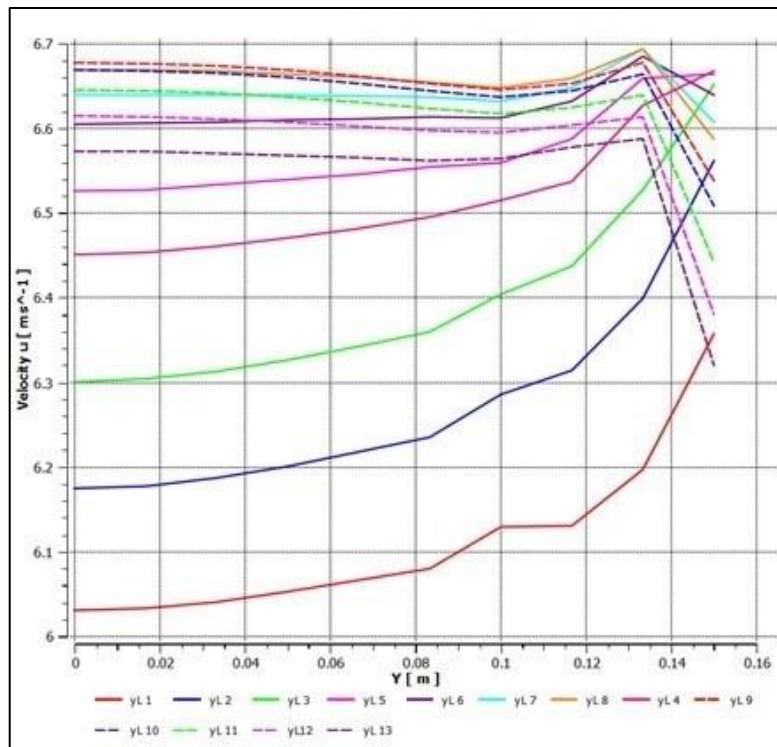


Figure 16: Linear Velocity distribution along the normal direction for flangeless diffuser in diffuser case.

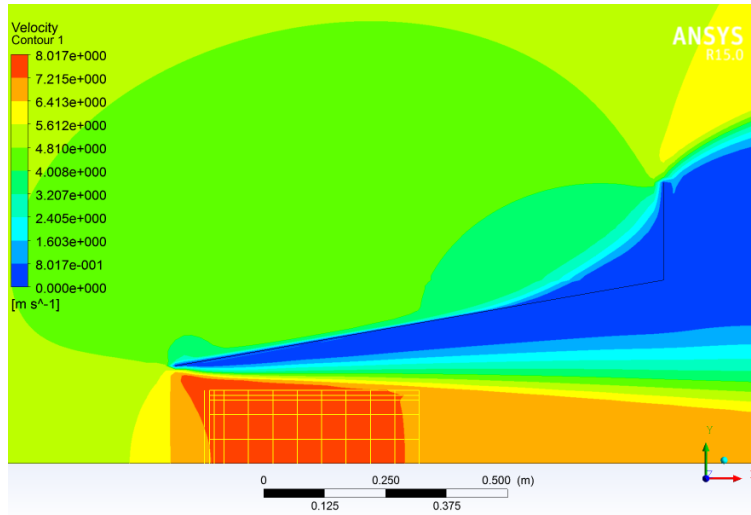


Figure 17: Linear Velocity Distribution results on Flanged Diffuser case

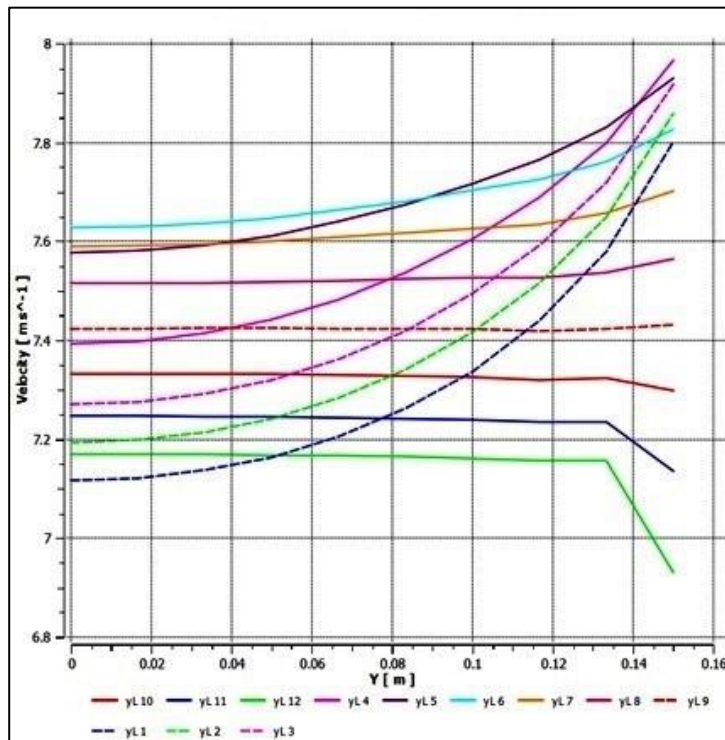


Figure 18: Linear Velocity distribution along the normal direction for flanged diffuser case.

The net wind speed augmentation is also found good, safe and high comparing with flangeless diffuser cases. Hence the optimal blade location for this case is 0.135 meter.

6. Conclusion

The DAWT is the miniaturization of a conventional bare wind turbine with the same output. But the wind turbines are limited with betslaw. As it avoids the so many accidents and added advantages this study is considered in the view of enhancing the diffuser design. The conventional flangeless diffuser is considered in comparing the performance of proposed designs. The flangeless

diffuser in diffuser design and flanged diffuser designs are proposed. The experimentations numerically investigated the wind speed augmentation. The results are classified and consolidated in the below table. It is clear that the proposed designs are safe. For the further research planned that optimizing the flange height for the flanged diffuser to maximize its performance. In case of the flangeless diffuser in diffuser case, it is planned that the parameter to be optimized to maximize its output.

References

1. Lewis, R.I.; Williams, J.E.; Abdelghaffar, M.A. 'A theory and experimental investigation of ducted wind turbines'. *Wind Eng.* 1977, 1, 104–125
2. Kumar and Sachendra. Development in Augmented Turbine Technology. *Int. J. Trend Res. Dev.*, vol. 4, no. 1, (2017), pp.2394–9333.
3. Y. Ohya, T. Karasudani, A. Sakurai, K. ichiAbe, and M. Inoue. Development of a shrouded wind turbine with a flanged diffuser, "J. Wind Eng. Ind. Aerodyn., vol. 96, no. 5, (2008), pp.524–539.
4. P. Schaffarczyk. Introduction to Wind Turbine Aerodynamics - Chapter 2. *Green Energy Technol.*, vol. 153, (2014), pp.7–21.
5. Kosasih and A. Tondelli, "Experimental study of shrouded micro-wind turbine. *Procedia Eng.*, vol. 49, (2012), pp.92–98.
6. R. F. Ghajar and E. A. Badr. An Experimental Study of a Collector and Diffuser System on a Small Demonstration Wind Turbine. *Int. J. Mech. Eng. Educ.*, vol. 36, no. 1, (2008), pp.58–68.
7. Matsushima, T.; Takagi, S.; and Muroyama, S. (2006). Characteristics of a highly efficient propeller type small wind turbine with a diffuser. *Renewable Energy*, 31(9), 1343–1354.
8. Wilson, R. E. and Lissaman, Peter B. S., "Applied Aerodynamics of Wind Powered Machines," Oregon State University, May 1974. 2. N.Y.U. Final Report on the ... 1974 MAY University . GI-418340 No. Grant Under State Oregon (RANN) Needs National to Applied Research Foundation, Science National the by Supported...
9. Jonkman, J M. Modeling of the UAE Wind Turbine for Refinement of FAST{ }AD. United States: N. p., 2003. Web.doi:10.2172/15005920.
10. Manwell et al., *Wind Energy Explained*, Wiley, Chichester, West Sussex, 2004
11. Emrah Kulunk, (2011), *Aerodynamics of Wind Turbines*, New Mexico Institute of Mining and Technology, USA
12. Balasem Abdulameer Jabbar Al-Quraishi, Nor Zelawati Binti Asmuin, Sofian Bin Mohd I, Wisam A. Abdal-wahid, Akmal Nizam Mohammed, Djamal Hissein Didane, "Review on Diffuser Augmented Wind Turbine (DAWT)", *International Journal of Integrated Engineering* Vol. 11 No. 1 (2019) 178-206.
13. T. Saravana Kannan, Saad A. Mutasher, Y.H. Kenny Lau, "Design and flow velocity simulation of diffuser, Augmented wind turbine using CFD", *Journal of Engineering Science and Technology*, Vol. 8, No. 4 (2013) 372–384.
14. Van Bussel, G.J.W. (2007). 'The Science of making more torque from wind: diffuser experiments and theory revisited'. *Journal of Physics: Conference Series*, 75, 1-1.